MARS PATHFINDER BATTERY PERFORMANCE

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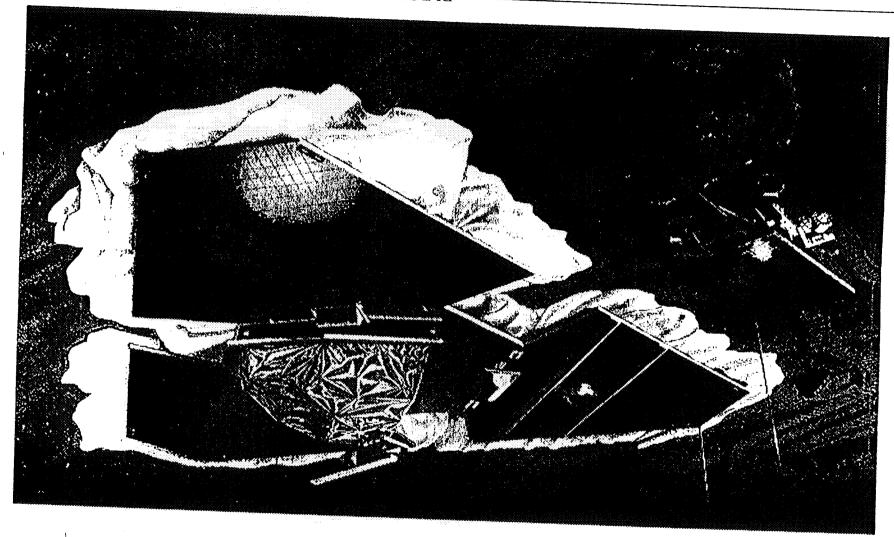
> NASA BATTERY WORKSHOP HUNTSVILLE, ALABAMA NOVEMBER 18-20, 1997

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OUTLINE

- MISSION REQUIREMENTS
- BATTERY DESIGN FEATURES
- BATTERY OPERATIONAL OVERVIEW
- BATTERY PERFORMANCE
 - PRELAUNCH
 - CRUISE
 - EDL
 - MARS OPERATIONS
- SUMMARY AND CONCLUSIONS

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MISSION REQUIREMENTS

•	PRE	LA	UNCH
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4 MONTH ACTIVE INVERTED STORAGE

LAUNCH

INVERTED BATTERY LAUNCH

CRUISE

7 MONTH ACTIVE STORAGE

• EDL

40 AH, 1080 WH

MARS OPR. CYCLES

30 CYCLES(1CYCLE/DAY, 8 HOUR CHARGE 16 HOUR DISCHARGE) 50% DOD

15 KG

WEIGHT

9.8" x 7.4" x 7.4"

DIMENSIONS

MPF BATTERY SPECIFICATIONS

VOLTAGE

27 V

CAPACITY

40 Ah

RATE CAPABILITY

1-5 A

PULSE CAPABILITY

40 A FOR 40 MSEC

CYCLE LIFE

40

WET LIFE

14 MONTHS

WEIGHT

15 KG

DIMENSIONS

9.8" x 7.4" x 7.4"

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KEY CELL DESIGN FEATURES

- Robust separator system 5 layers of cellophane and 2 layers of polymer membrane.
 To achieve long calendar and cycle life
- Triple redundant case-to-cover seal including basic ultrasonic seal.
 To prevent electrolyte leakage
 To improve safety
- Large cell plate area approximately 200 square inches.
 Low temperature operation
 For enhanced pyro-firing
- Unique leak-free cell vent valves.
 For inverted battery operation allow gas venting under off-limit operation
- Minimal free electrolyte
 For inverted battery operation

KEY BATTERY DESIGN & PROCESS FEATURES

- Titanium fabricated battery case with sealable cover and gasket.
 Light weight construction
 gas and electrolyte containment
- Battery vent valve and pressurization port.
 Redundant valve to protect cells from electrolyte loss
- Battery heater and two temperature sensors.
 Thermal management for charge control
- Over-pot of cells, surface conformal coating, connector back side potting and electrolyte absorption system.

Prevent ionic conductive paths Improve safety

•Cell Matching and selection Extend cycle life

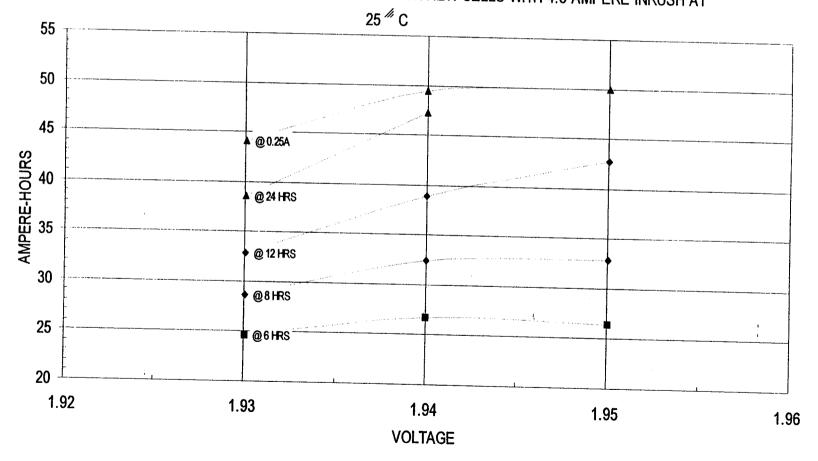
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KEY BATTERY OPERATING STRATEGIES

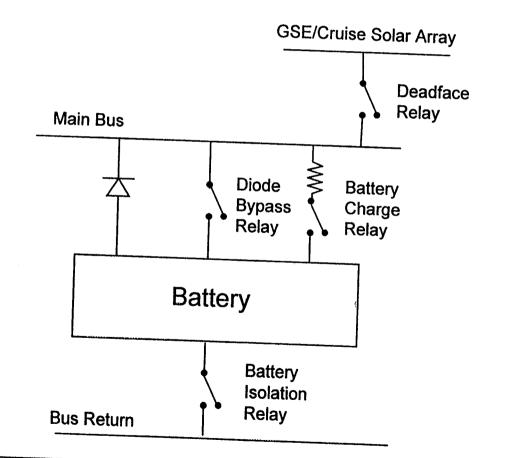
- Battery was mounted inverted and maintained at 12 C during pre-launch and launch Phases
- •Battery was partially discharged to 80% SOC and on open circuit stand, at -5 to 0 degrees C, during 7 Month cruise period.
- Battery was charged at end of cruise through 1.2 Ohm resistor to 0.2 A cut-off and 1.95 V / cell Ave.
- Battery was heated to 15 to 20 degrees C prior to all charges.
- Battery was charged during Mars operations without resistor to a selective shunt limiter controlled maximum voltage. Six voltage settings were available with 1.95 V / cell Ave.., the nominal full charge selection.
- Battery was taper charged to a constant shunt limiter voltage, 1.95 V / cell Ave.. well below the 2 V. limit employed in constant current charging with same full charge result.

BATTERY CAPACITY VS CHARGE VOLTAGE

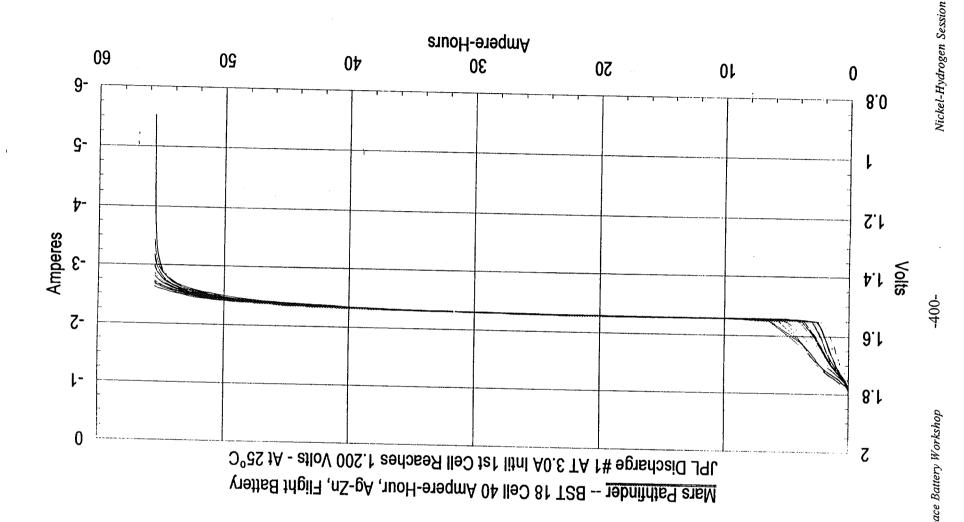
MARS PATHFINDER -- 16 BST Ag-Zn CELLS WITH 2+5 SEPARATOR SYSTEM CONSTANT VOLTAGE CHARGE CHARACTERISTIC FOR NEW CELLS WITH 4.5 AMPERE INRUSH AT



MARS Pathfinder Power System Configuration Relays

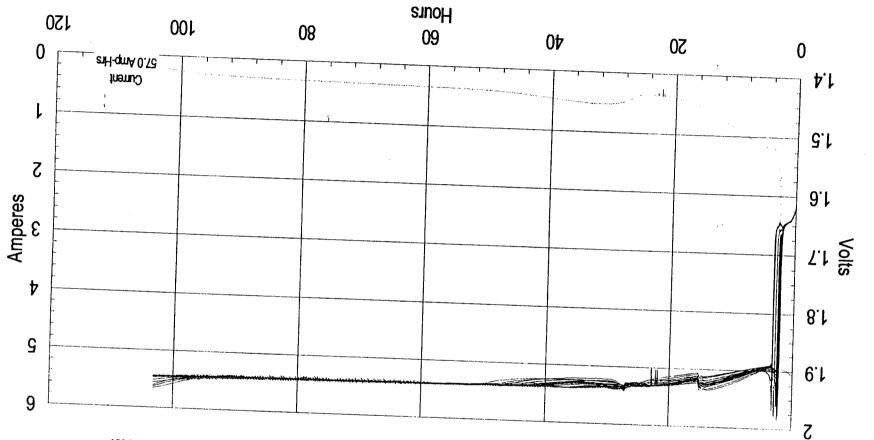


MPF BATTERY PRELAUNCH DISCHARGE



MPF BATTERY PRELAUNCH CHARGE

Mars Pathfinder -- BST 18 Cell, 40 Ampere-Hour, Ag-Zn, Flight Battery JPL Charge #2 AT 1.960 Volts Per Cell To A 0.2A Cut Off At 25°C - With 1.14 Ohm Resistor



94

04

Z# 184= # 184 **-**

- Battery Voltage

- Battery Current

Day of Mission

90

TEMPERATURE DURING CRUISE

GNA 30ATJOV YASTTA8 39M

gg

09

99

01-

G-

0

G

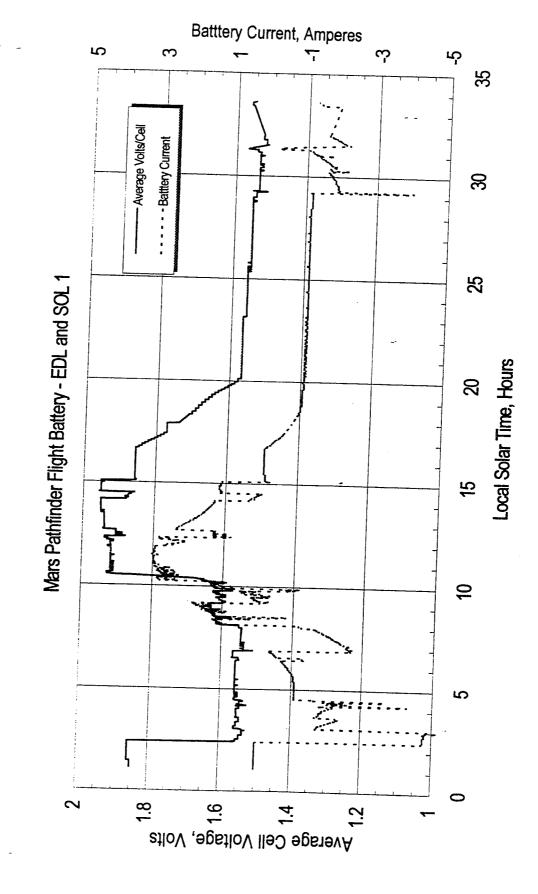
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Degrees Celsius

Nickel-Hydrogen Session

32

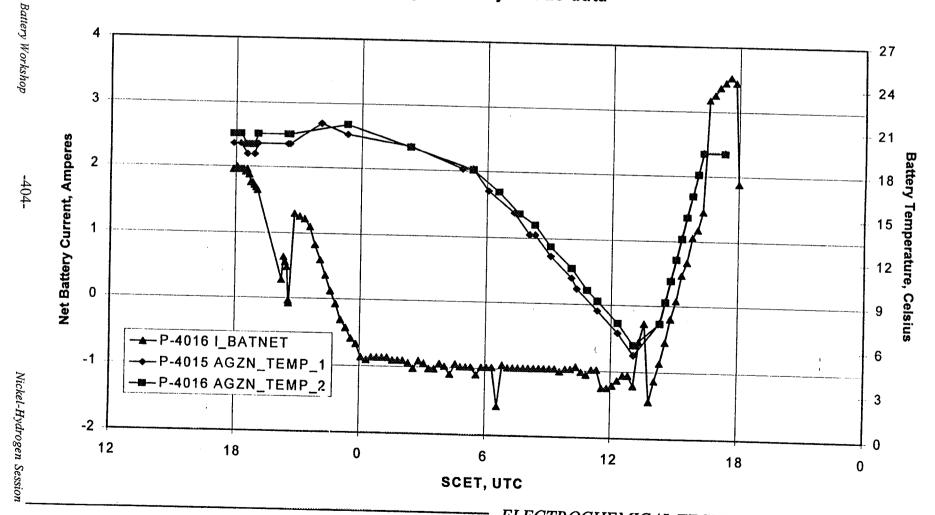
MPF BATTERY PERFORMANCE-EDL/SOL1



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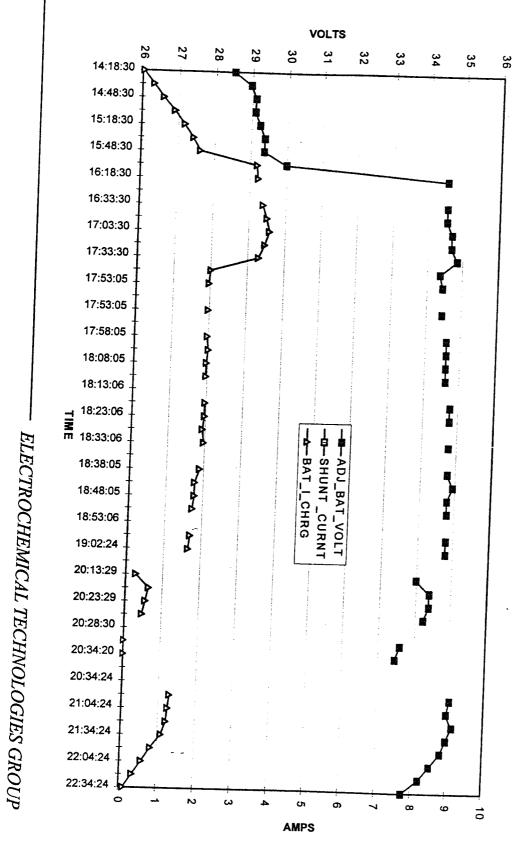
MPF BATTERY PERFORMANCE- SOL25

Flight Battery Sol 25 data



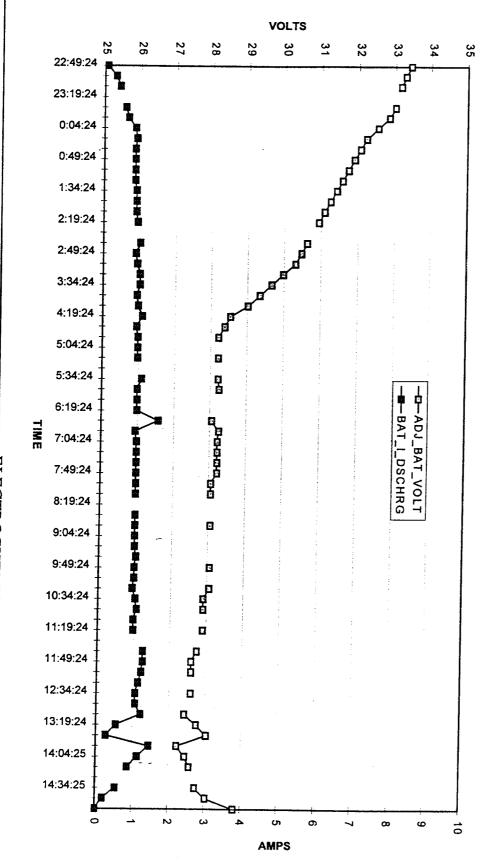
MPF BATTERY PERFORMANCE SOL-25

SOL #25 CHARGE



MPF BATTERY PERFORMANCE-SOL 25

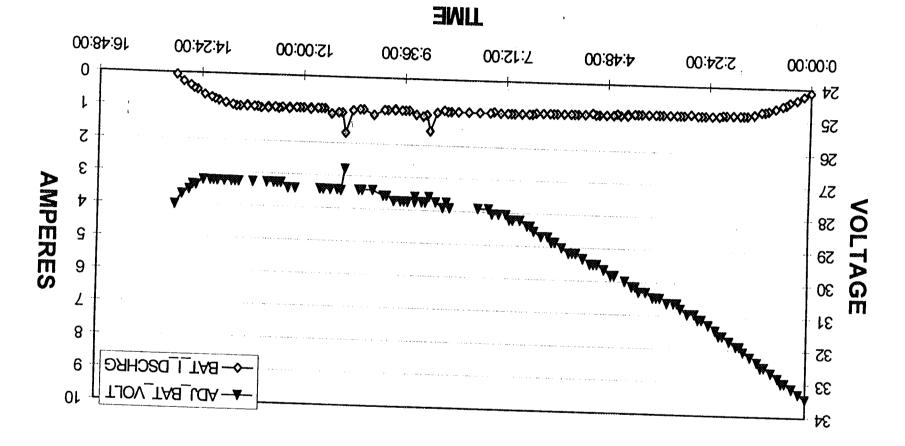




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MPF BATTERY PERFORMANCE-SOL 68

SOL 68 DISCHARGE



SUMMARY AND CONCLUSIONS

- The first use of a silver-zinc battery in a spacecraft application that called for extensive rechargeable operation after 12 months of active stand hasproven to be very successful.
- BST has developed a silver-zinc battery with unique design features for the Mars Pathfinder mission.
- JPL has developed battery management strategies to meet the Mars Pathfinder mission requirements.
 - Partial SOC at low temperature and open circuit stand was found to be the most effective method for insuring extensive cycle life following a long period of active storage
 - Silver-zinc battery charging at a reduced constant voltage was shown to provide full charge and a capability of supporting a large number of cycles.
 - Launch of an inverted silver-zinc battery was shown to be possible when a leak free cell vent valve is employed.
 - The use of a silver-zinc battery and shunt limiter in a direct energy transfer power system resulted in a very energy efficient design approach.



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